

SUPPLEMENTAL TYPE CERTIFICATE

10059003 REV. 1

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

CERTIFYNATION, LLC

**N53W24747 S CORPORATE CIR
SUSSEX WI 53089-4376
USA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: EASA.IM.A.120

Type Certificate Holder: THE BOEING COMPANY

Type: 737

Model: 737-600/ 737-700/ 737-800

737-900/ 737-900ER

Original STC Number: FAA STC ST04040CH

Description of Design Change:

Activation of Smart Runway and Smart Landing (SRSL) Runway Awareness Advisory System (RAAS) Functions for Honeywell Mark V and Mark V-A Enhanced Ground Proximity Warning Systems (EGPWS).

EASA Certification Basis:

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements for the following paragraph(s) at a later amendment:
CS25 Amdt.2 for affected area.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Cologne, Germany, 20 November 2017



Carla TORIO
Supplemental Type Certificates & Special Projects
Section Manager



Associated Technical Documentation:

Master Data List, Ref. CN15-005MDL-00, Revision 4.

Airplane Flight Manual Supplement, Ref. CN15-005AFM-02, Revision 6

or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)" and/ or the Technical Implementation Procedures of EU/ USA Bilateral Agreement.

Limitations/Conditions:

As per:

- Master Data List, Ref. CN15-005MDL-00 Revision 4 (Section 1.4).
- Airplane Flight Manual Supplement, Ref. CN15-005AFM-02 Revision 6 (Section II).

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -